



**CASR**

FAA Center for Aviation Systems Reliability

# Engineering Studies of Fluorescent Penetrant Inspection

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Defense Working Group  
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<http://www.cnde.iastate.edu/faa-casr/fpi/index.html>



- 1999 – 2002 – Cleaning and Drying Studies performed as part of the Engine Titanium Consortium
- 2002 – 2006 – Engineering Assessment of Fluorescent Penetrant Inspection performed as part of Center for Aviation Systems Reliability effort



- Provide engineering data to support decisions regarding the safe application and relevant use of FPI
- Includes data to support changes in specifications
- Generate tools for use by airlines and OEMS that improve FPI processes
- Strong industry team with extensive experience



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IOWA STATE UNIVERSITY  
OF SCIENCE AND TECHNOLOGY



**Pratt & Whitney**

A United Technologies Company



**Rolls-Royce**

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*Cooperative university/industry program which brings together aircraft and engine OEMs, airlines, vendors, as well as technical expertise from the NDE community.*

**ISU: Lisa Brasche, Rick Lopez, Dave Eisenmann, Bill Meeker**  
**FAA: Al Broz, Paul Swindell, Dave Galella**

## Industrial Advisory Panel

Boeing - Long Beach

Dwight Wilson, John Petty

Boeing - Seattle

Steve Younker, Mike Davis

Delta Airlines - Atlanta

Lee Clements

United Airlines - Indianapolis

Dave Arms, Bob Stevens

Pratt & Whitney - EH and WPB

Kevin Smith, John Lively

Rolls Royce - Indianapolis and Darby

Pramod Khandelwal, Keith Griffiths,

Bill Griffiths, Tom Dreher

GE Aircraft Engines

Terry Kessler, Thadd Patton, Wayne

Kitchen, Phil Keown

Sherwin - Cincinnati

Sam Robinson

D&W Enterprises - Denver

Ward Rummel



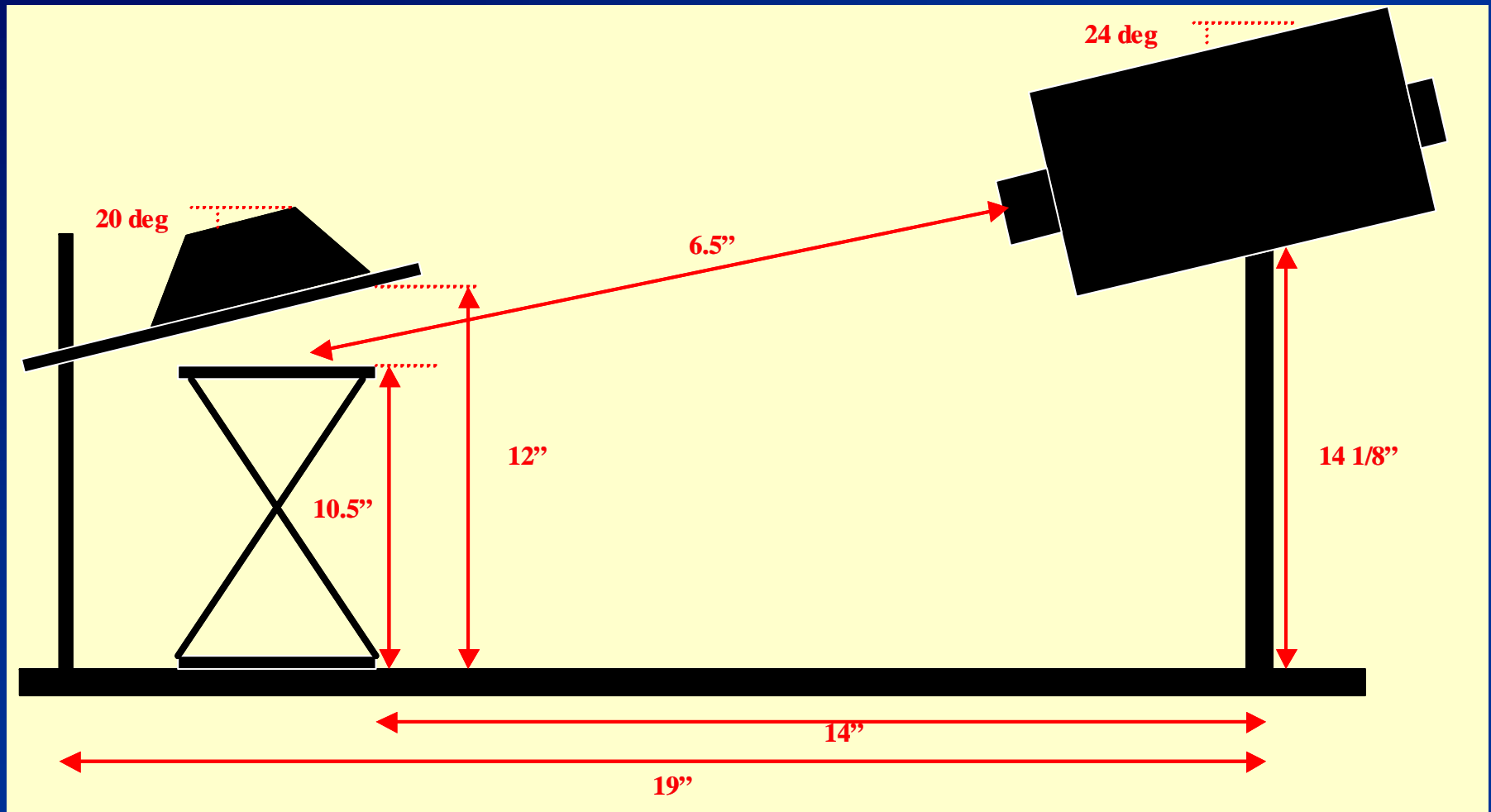
- Define factors for which engineering data is deficient
  - Change in process, e.g., environmental changes
  - Change in applications
  - Data not available in the public domain
- Design engineering study that provides quantitative assessment of performance
  - Brightness measurements
  - Digital recording of UVA indication
  - Probability of Detection
- Complete study using either lab or shop facilities as appropriate
- Distribute results through use of web
- Support changes to industry specifications as warranted
- Utilize results to update/create guidance materials
- Transition process to airlines for internal, self-assessment



- Used rigid fixturing to assure repeatability with transportability for brightness measurements
- Photo Research PR-880 Photometer used to record indication brightness in ft-Lamberts









- Requires access to typical drying, cleaning and FPI methods used in commercial aviation
- Several partners have provided access to their facilities
  - Access to cleaning lines for Ti and Ni as well as mechanical blasting facilities
  - FPI line for sample processing
  - Inspection booth for characterization and brightness measurements







- 15 - 20 samples per basket
- 20 minute penetrant dwell
- 90 second pre-wash
- 120 seconds emulsifier contact with vertical motion
- Two 30 second cycles of air agitated water rinse, then a 90 second post-wash





- Samples dried for 8 minutes at 150°F
- Drag-through application of developer
- 10 minute development time
- Brightness reading using Spotmeter
- Length reading using UVA and image analysis software

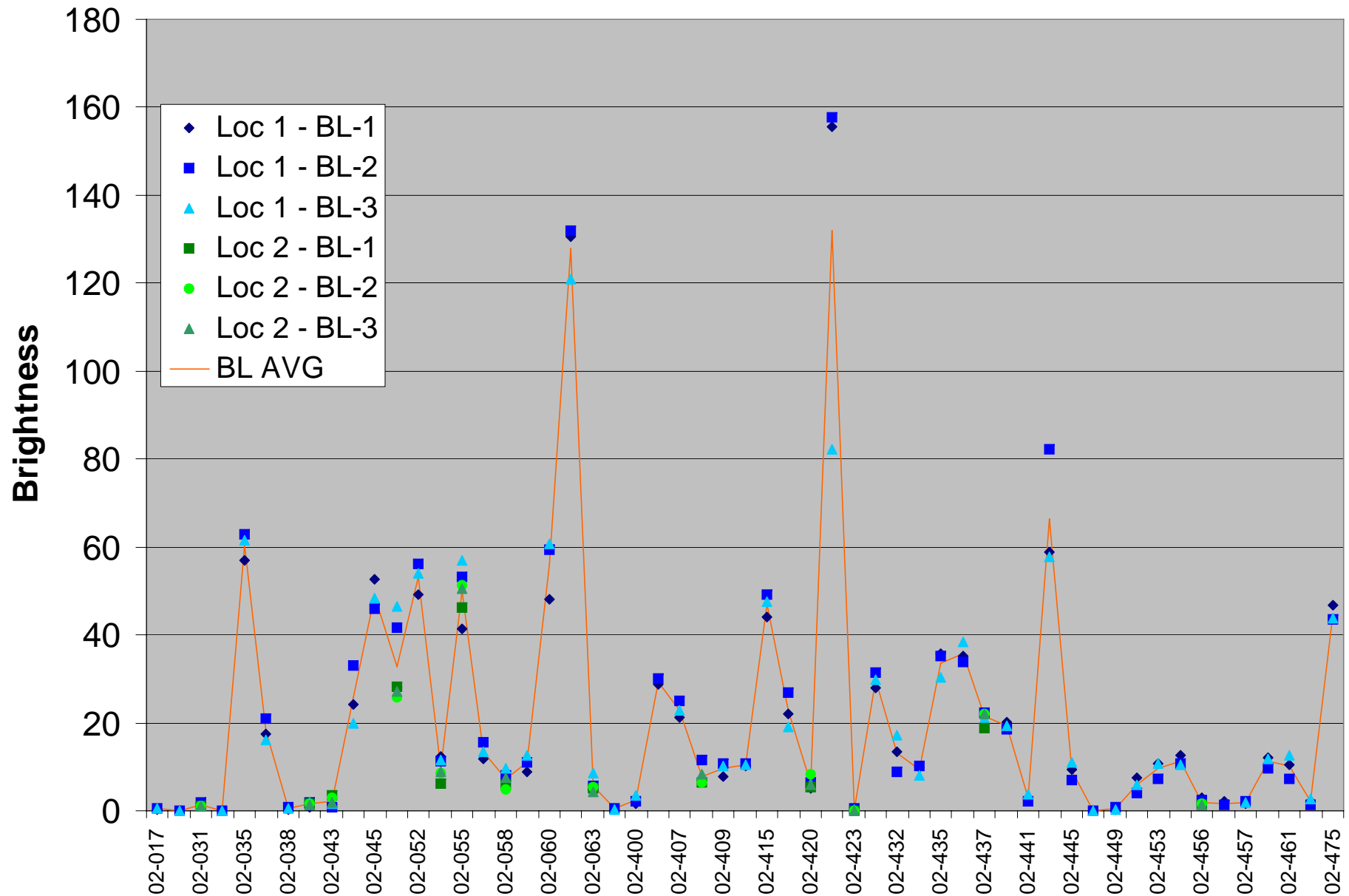
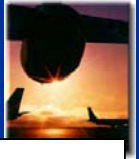




- Topics for engineering studies selected and prioritized by team
  - Subteams developed for experimental design with review by the full team
  - Experimental efforts to take place at various industry locations
  - Definition currently underway
- ES – 1 – Developer Studies
  - ES – 2 – Cleaning Studies for Ti, Ni and Al
  - ES – 3 – Stress Studies
  - ES – 4 – Assessment tool for dryness and cleanliness
  - ES – 5 – Effect of surface treatments on detectability
  - ES – 6 – Light level Studies
  - ES – 7 – Detectability Studies
  - ES – 8 – Study of Prewash and Emulsification Parameters
  - ES – 9 – Evaluation of Drying Temperatures
  - ES – 10 – Part geometry effects
  - ES – 11 – Penetrant Application Studies
  - ES – 12 – Relationship of part thickness to drying method



- Utilized standard sample process with baseline established using dip/drag method of developer application
- Evaluated four developer chambers and wand application methods at two locations
- Same penetrant process (level 4 PE) and chemistry use through out





- Chamber a – Developer applied through linear diffuser located at top and bottom of chamber
- Chamber b – Developer applied from circular diffuser located at top and bottom of chamber
- Chamber c – Developer applied from circular diffuser located at top of chamber
- Chamber d – Developer applied from two nozzle diffusers located at bottom of chamber
- Manual spray – Low pressure, high volume manual application
- Dip/drag – Hand application of individual samples. Used for baseline measurements.



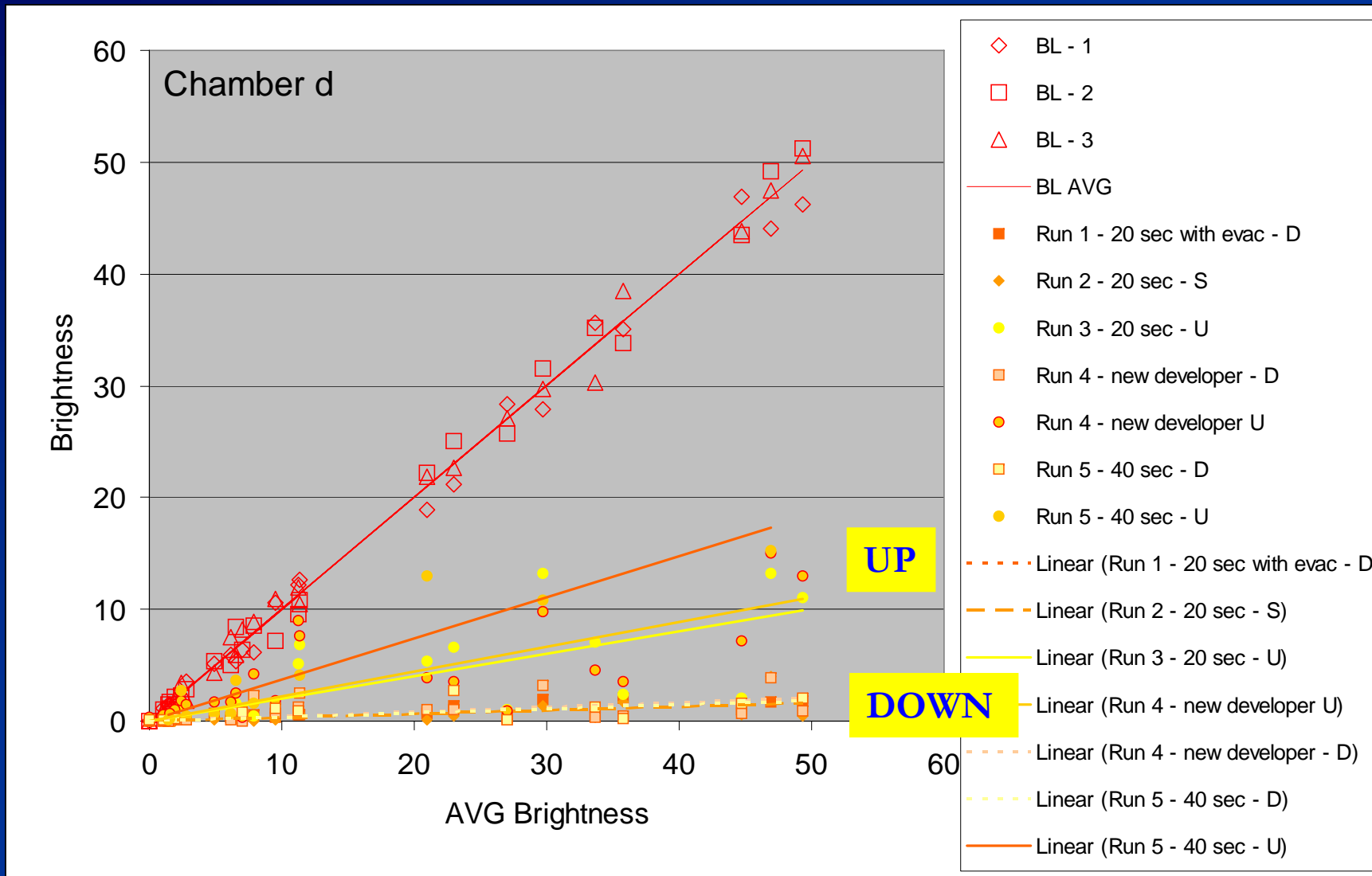
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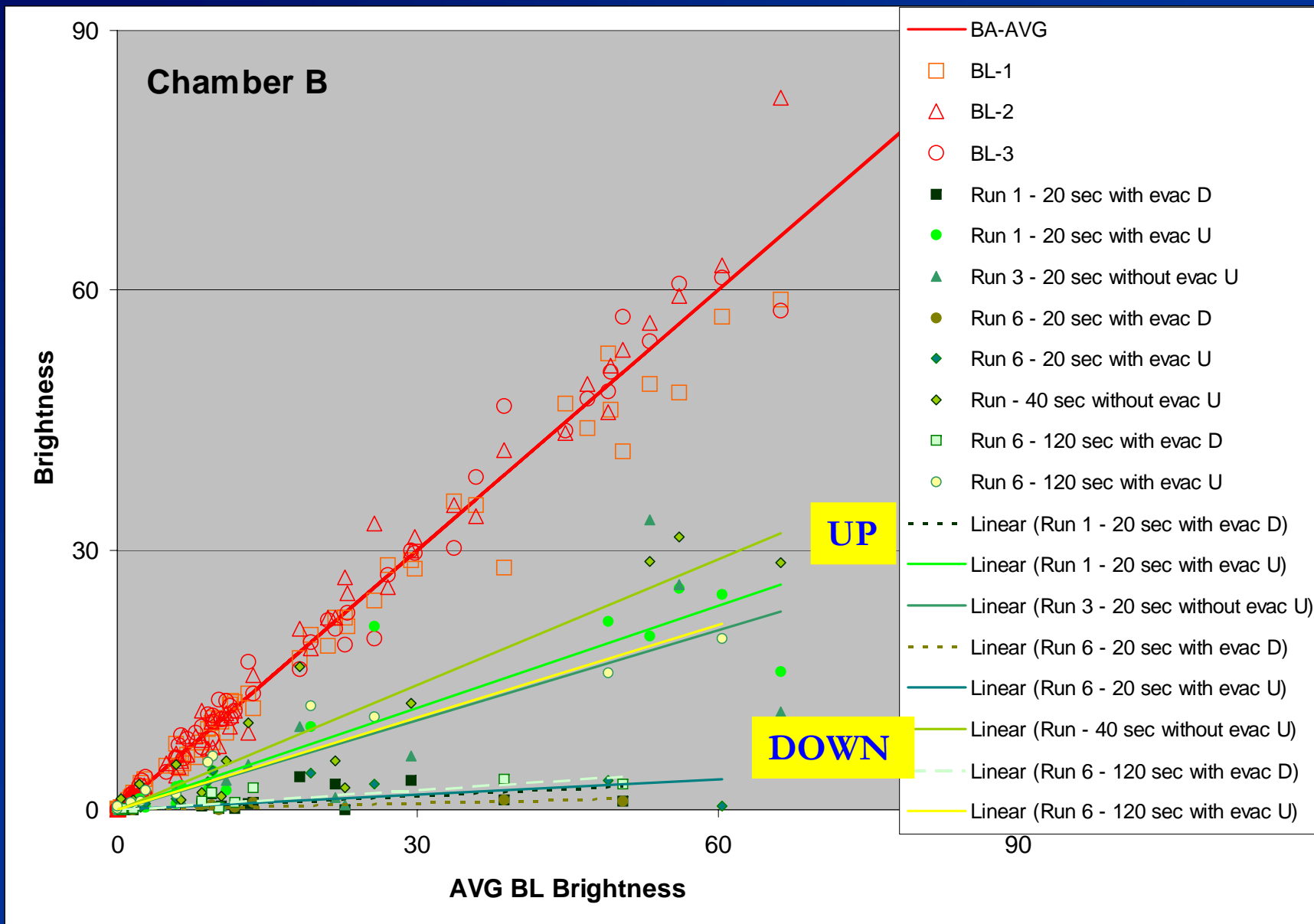
## Chamber D Characterization



- Chamber contains two jets, at approximately  $\frac{1}{4}$  and  $\frac{3}{4}$  of the chamber length
- Jets located below rollers
- Typical operation of 5 sec developer application followed by 10 min dwell in chamber









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## Manual Spray Application

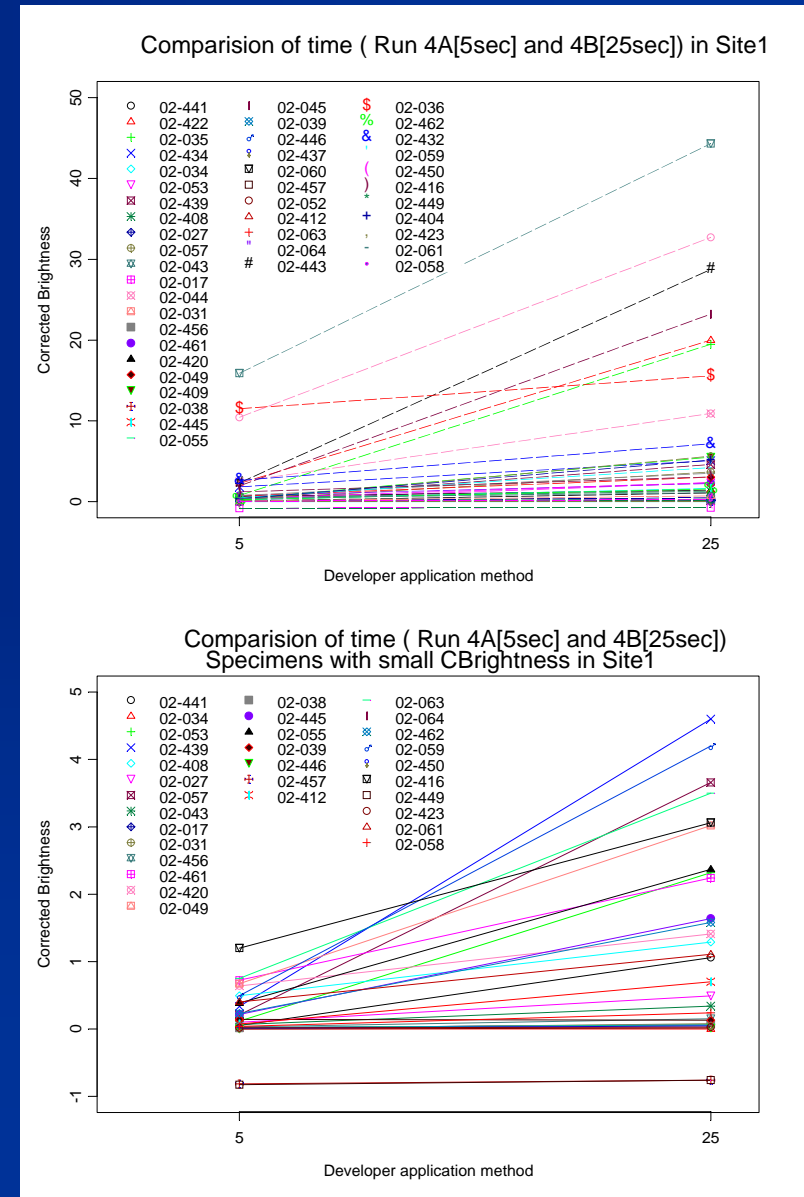


- Low pressure, high volume spray
- 5 and 25 sec runs completed using lobster cage with cracks in D, S or U position
- 60 and 120 sec runs completed with samples all in U position





- Increasing time of manual spray application from 5 to 25 sec showed significant improvements in brightness
- Emphasize as part of future training opportunities



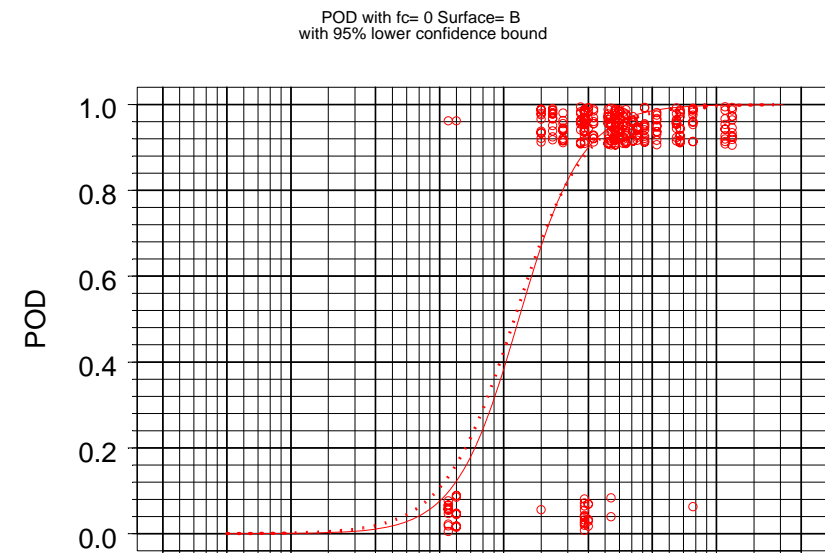
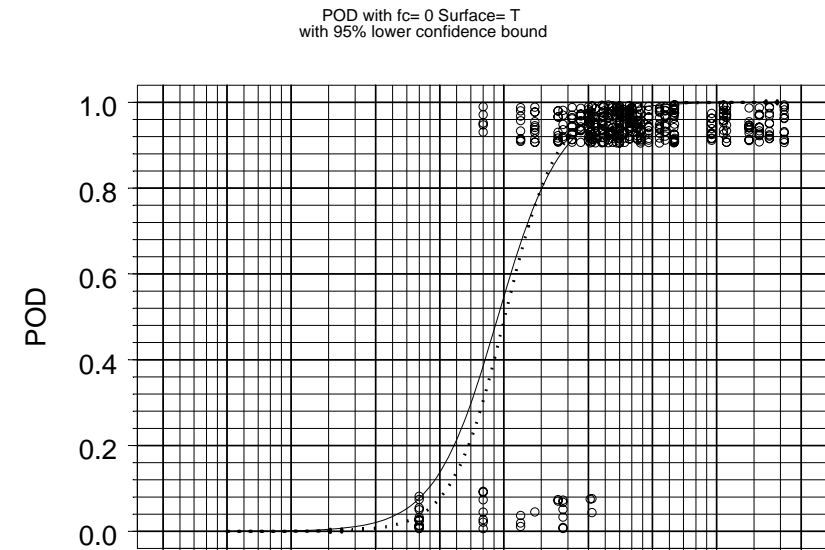


- Statistical analysis showed:
  - Differences were found in location within the chambers
    - Right/left effects in Chamber B but not Chamber A for cracks in up position
    - Improved brightness in middle of Chamber B compared to either end for cracks in up position
    - More variation at front of Chamber D than middle and back of chamber
    - No right/left, front/back or level effects for cracks in down position
    - No level (top, middle bottom) effect found in Chamber A, B or D
  - Most significant effect was crack orientation (up, down, sideways)



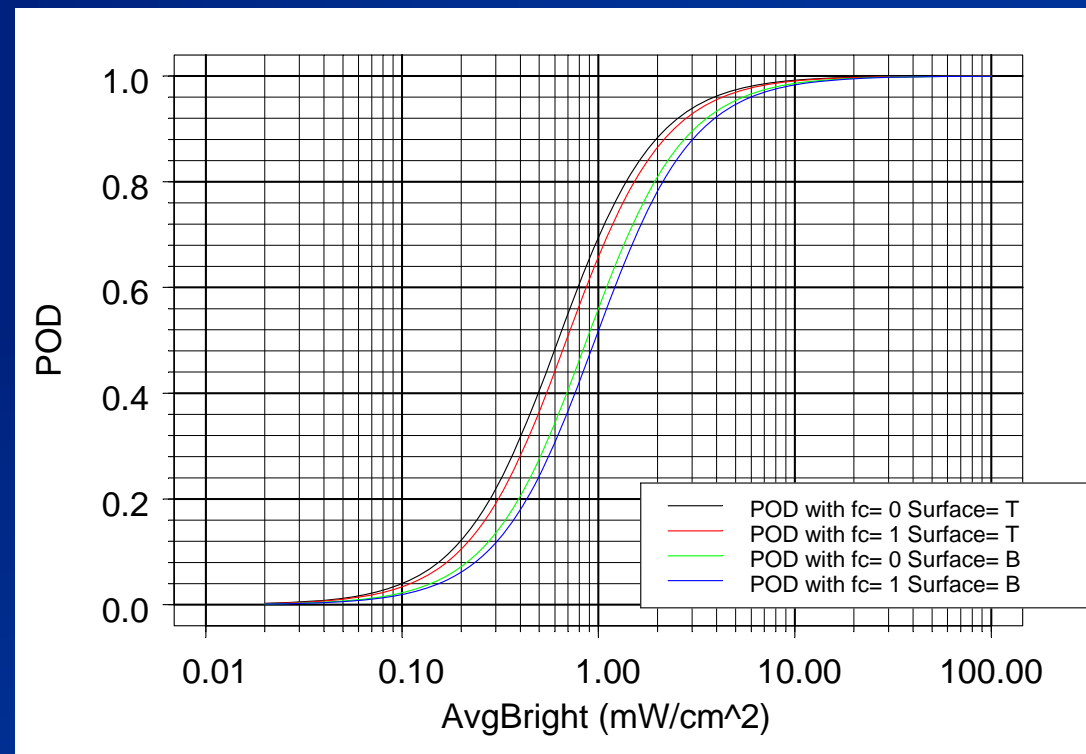


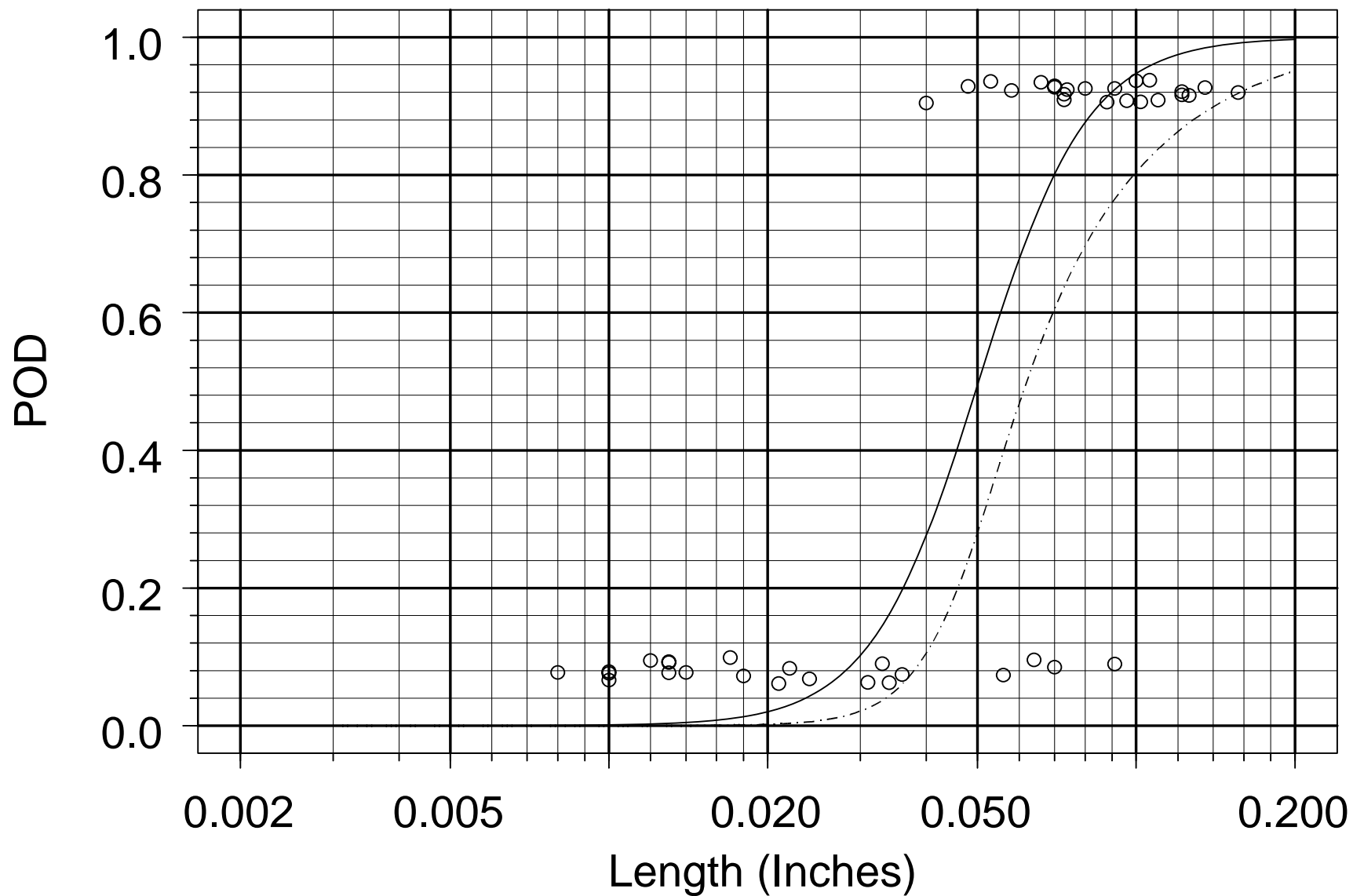
- Completed POD study which correlates brightness to detectability
- Used two sample sets, two inspectors under multiple UV intensity level, white light level combinations
- Evaluated indication location (top or bottom) of panel
- Significant differences can occur





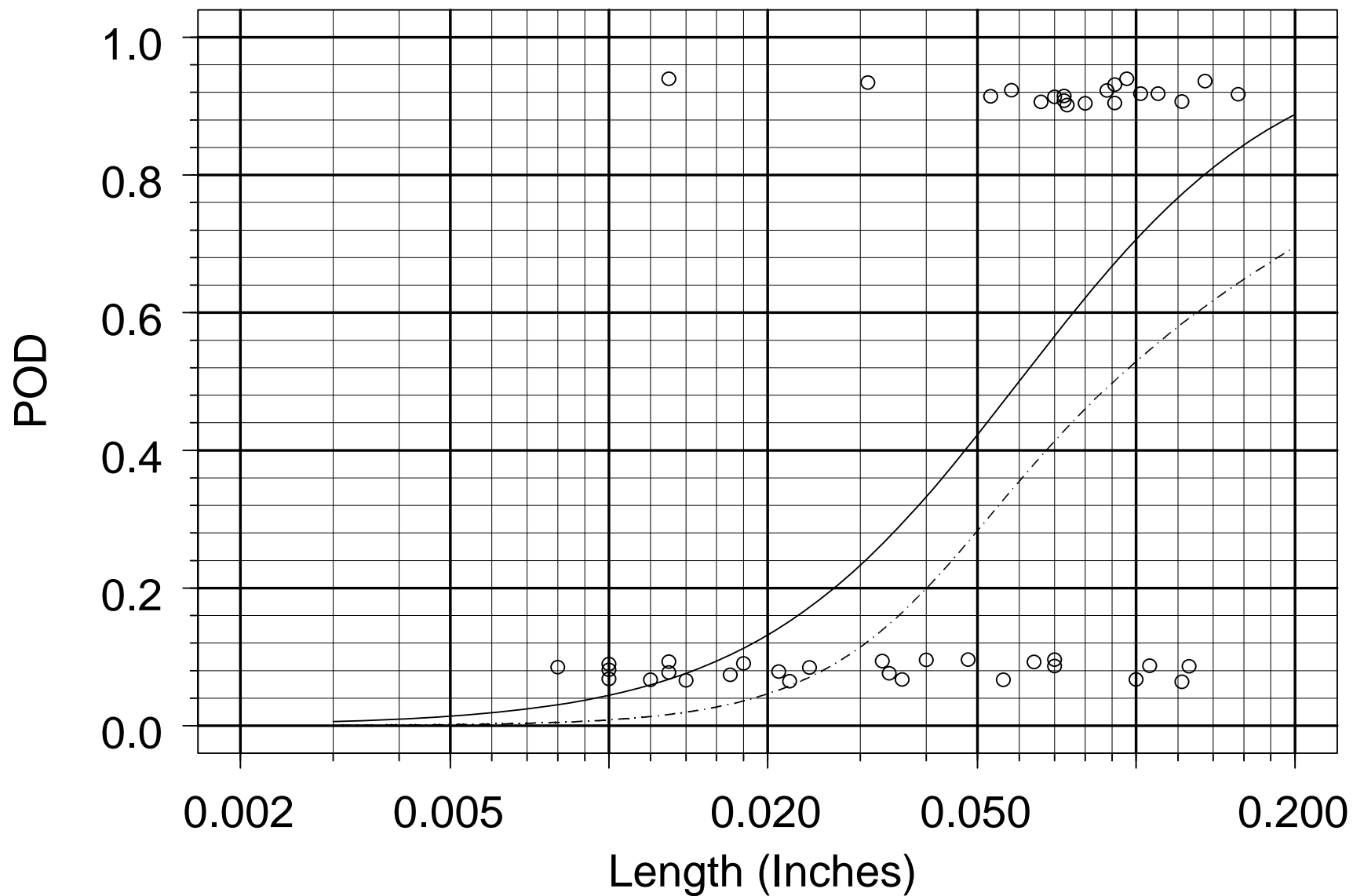
- POD is correlated to brightness
- UVA intensity of 5000  $\mu\text{watts}/\text{cm}^2$  lead to ~15 mil improvement in POD when compared to 1000 and 3000  $\mu\text{watts}/\text{cm}^2$
- Increasing whitelight contamination led to significant reductions in POD in excess of 100 mils







R4.I2.DevCh.5kuva.0fc  
Hit-Miss POD with 95% lower confidence bound





- Developer application is critical to overall FPI performance
- Developer application by dip/drag yields brighter indication than with any of the developer chamber or wand application methods
- No indications were “lost” but detectability improves with brightness – optimal process will yield bright indications
- Sample orientation matters
  - Avoid barriers that prevent direct application of the developer
  - Ensure chamber configuration or part handling fixtures (rollers, baskets, etc.) don’t hamper application
  - No metal-to-metal contact
  - May require multiple trips through the chamber to ensure adequate coverage on all surfaces
- White light contamination matters



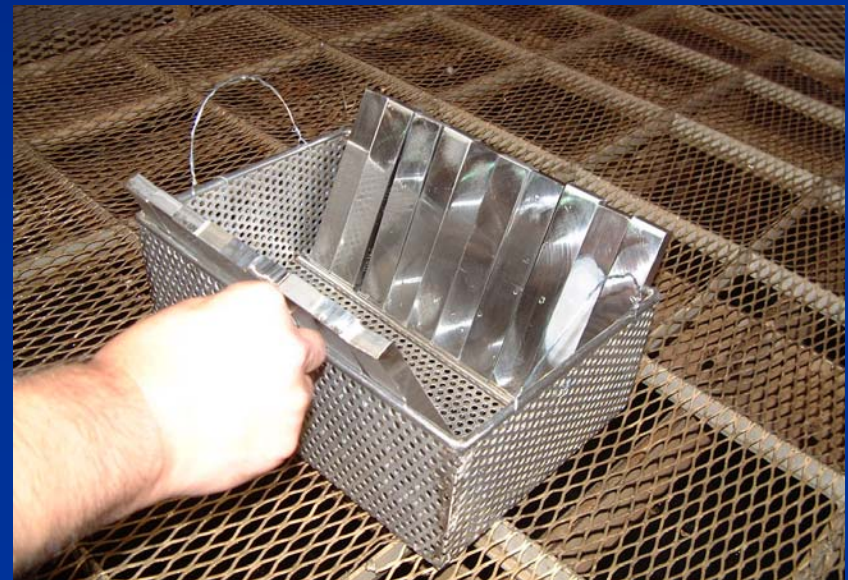
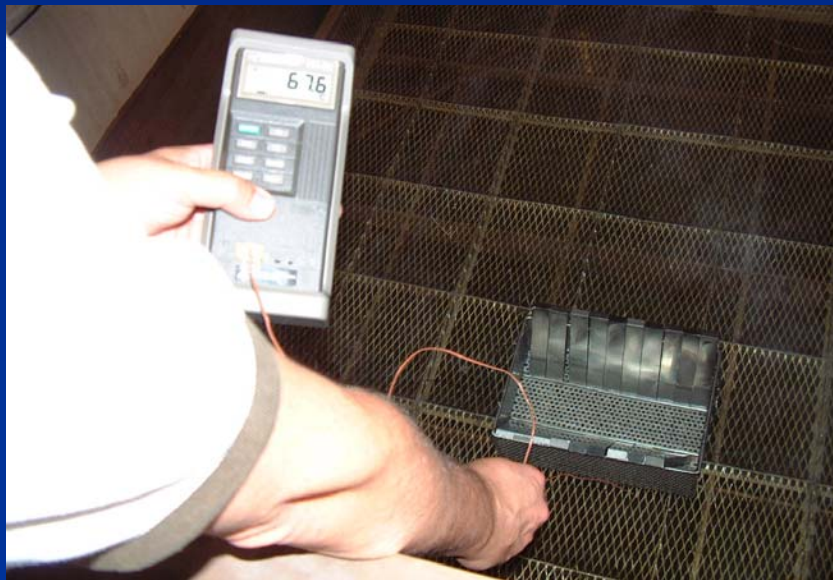
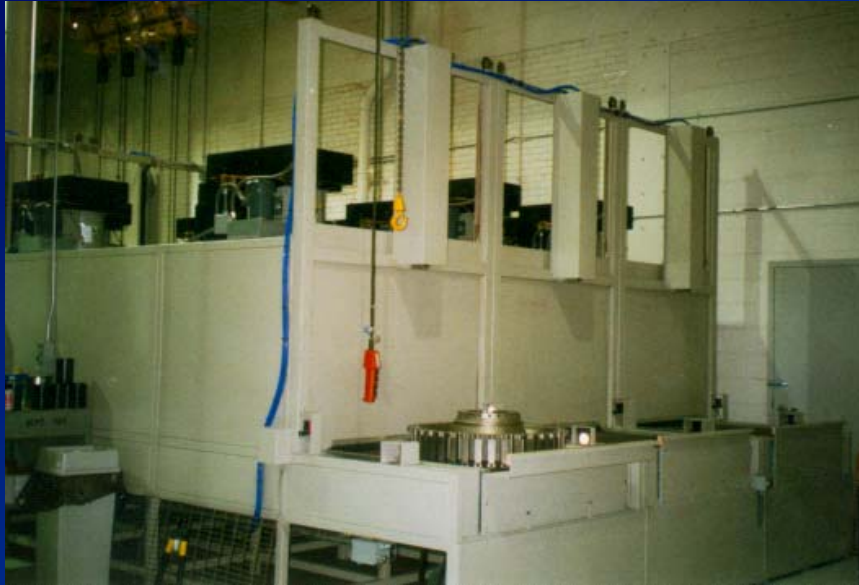
## Drying study parameters

- Ultrasonic acetone clean 30 minutes
- Flash dry
  - Water bath at RT (82F – 28C)
  - Flash dry at 150F (66C)
- Oven dry
  - Water bath at RT (82F – 28C)
  - Oven dry at 225F (107C) for 30 minutes
- FPI Process
  - Cool to 40C prior to FPI
  - ZL-37 – UltraHigh Sensitivity Post Emulsified Penetrant
- Spotmeter brightness and digital recording of image



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## Drying Study







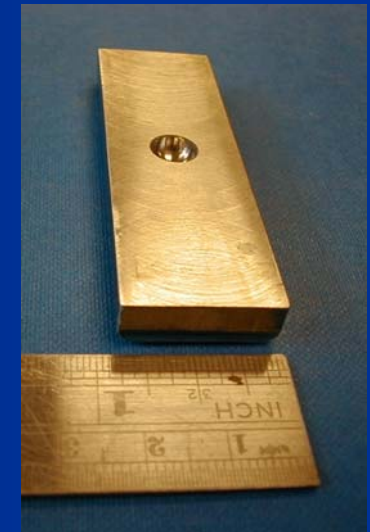
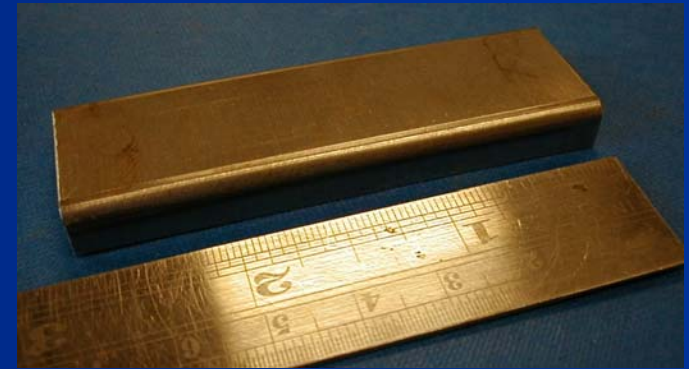


- Statistical analysis of brightness and UVA lengths did not reveal significant differences between the two drying methods at the temperatures used in this study, i.e., flash drying at 150°F and oven drying at 225°F
- Potential factors not considered in the current study are the effect of thermal mass, potential differences in penetrant level, and a range of drying temperatures. Additional studies that explore these factors are underway.
- While significant differences were not found between the two methods, the importance of process monitoring and control for either method should be emphasized in specifications, standard practice documents, and training/guidance materials. Without careful adherence to the recommended practices, reductions in detectability can occur with either method.
- A comparison of the results of quantitative brightness measurements such as completed in this program and the more traditional POD study is underway



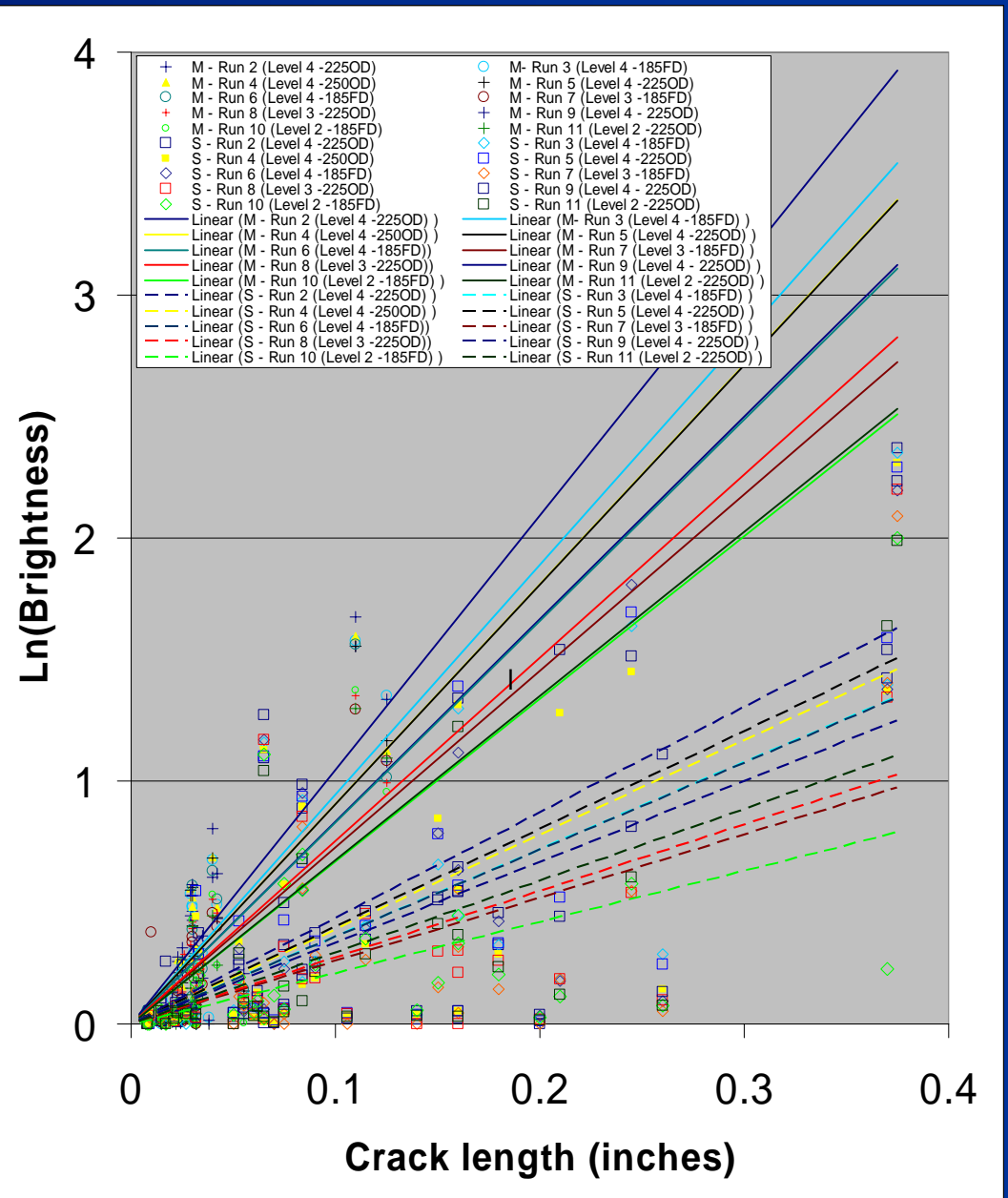


- Samples included shot peened and as machined surfaces
- Penetrants
  - Level 4 ultrahigh postemulsifiable: Magnaflux ZL – 37
  - Level 3 surfactant based water wash: Magnaflux ZL – 67
  - Level 2 oil based water wash: Magnaflux ZL – 60D
- Additional drying parameters
- POD data generated





- Results analyzed as function of penetrant method, drying parameter, and surface finish
- Strongest factor was surface finish
- Expected differences found between penetrant levels







- For sample size and crack size used, differences were not found between the two drying methods. Factors not considered include thermal mass which will be accessed as part of future studies using real parts and appropriate fixtures.
- Differences were found between the two surface finish conditions. Detectability in shot peened surfaces present on these samples was lower than machined surfaces.
- Differences were found between penetrant method with Level 4 found to be more sensitive than Levels 3 or 2. Differences between levels 2 and 3 were not significant for the rinse times used in this study.



- Evaluate geometry and high thermal mass effects on brightness in response to changes in processing parameters.
- Utilized real part with fatigue cracks generated during spin pit test and provided for use by Rolls Royce.
  - Weights approx. 300 lbs
  - Waspaloy material
  - Changing geometry
  - High mass to volume ratio
  - Shot peened surface

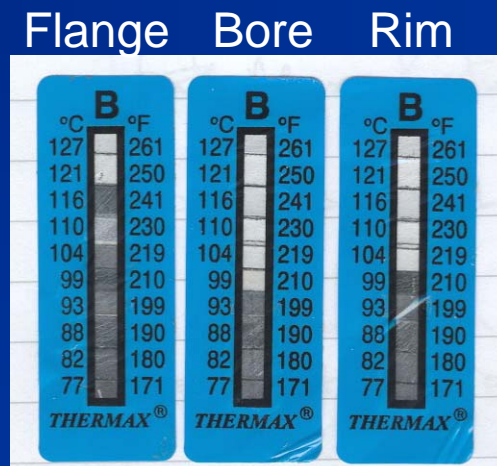
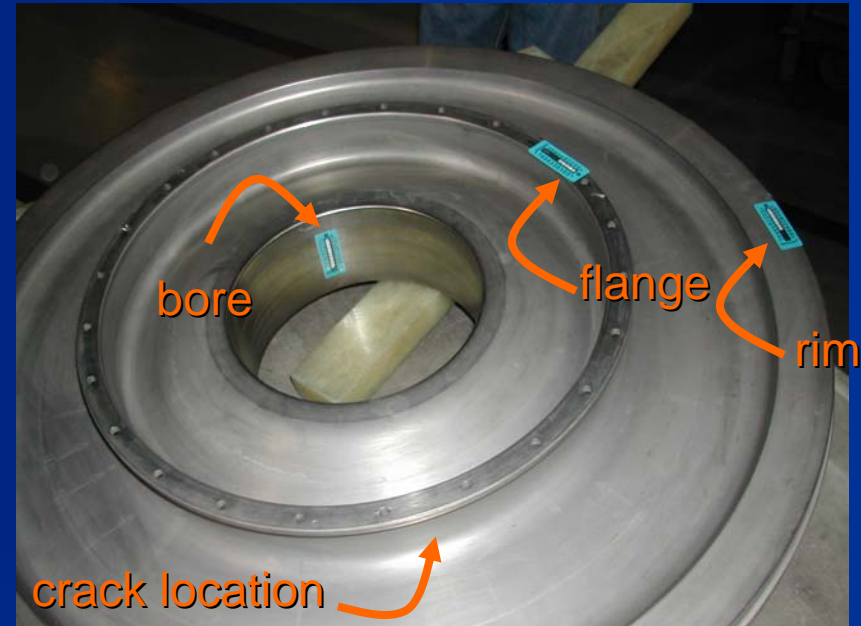


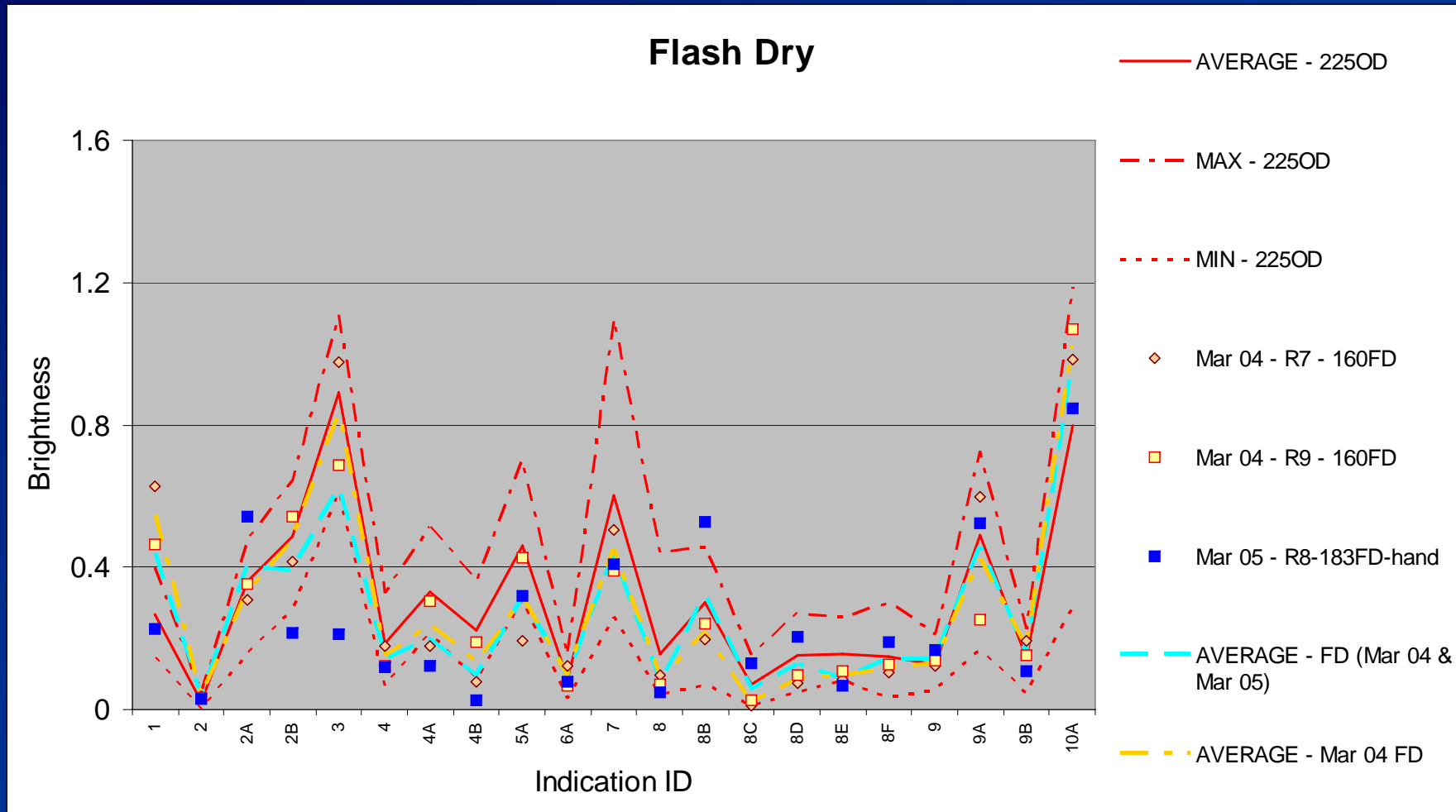


- Vapor degrease 5 mins @ 183F (This increased to default of 20 mins due to persistent FPI indications).
- Oven dry @ 225F for 30 mins
- Cool to 104F (forced air cooled using fan)
- Level 4, PE penetrant (ZL-37), dipped and dwelled for 20 mins
- Wash 60 seconds
- Emulsify using ZR-10B with agitation for 120 seconds
- Wash 60 seconds
- Pre developer dry @ 160F for 20 minutes
- Dry powder developer, ZP-4B, hand processed 10 minute dwell



- Temperature gages used to determine variation with part geometry
- Order of increasing temperature:
  - Inner (bore)
  - Outer (rim)
  - Middle (flange)





- Similar average brightness between FD and OD in Mar 04
- More variability and lower average brightness found with FD in Mar 05 than 225OD, possibly due to emulsifier effects



- Average brightness similar for both oven dry temperatures, i.e., 225F and 250F
- FD data requires additional statistical analysis
  - More variation found with FD when compared to 225OD, original emulsifier
  - With new emulsifier, FD and OD performed similarly
- Dust chamber application shows similar brightness debits to those found using lcf samples
- Use of heavy duty alkaline clean led to improvements in brightness
- Recommend final study to establish minimum acceptable drying temperature for parts, i.e., energy savings benefits



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## More information



- Website to provide background info and publish technical results
- Link to FAA Reports available

FPI homepage - Netscape

File Edit View Go Communicator Help

Back Forward Reload Home Search Netscape Print Security Shop Stop

Bookmarks Location: <http://www.cnde.iastate.edu/faa-casr/fpi/index.html>

Instant Message WebMail Contact People Yellow Pages Download Channels

**CASR**  
FAA Center for Aviation Systems Reliability

**Engineering Assessment of  
Fluorescent Penetrant Inspection**

- The Center for Aviation Systems Reliability (CASR) was established in 1990 to provide results that address the inspection needs of commercial aviation and lead to safety improvements through their implementation.
- In September 2002, CASR partnered with industry to begin a new research program entitled Engineering Assessment of Fluorescent Penetrant Inspection.
- This website was established to document the results of this and other Federal Aviation Administration (FAA) -funded programs working to improve the reliability of the Fluorescent Penetrant Inspection (FPI) process.

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